# EL3210 Multivariable Feedback Control

#### Lecture 8: Youla parametrization, LMIs, Model Reduction and Summary [Ch. 11-12]

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# Todays program

Optimal control problems

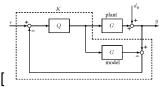
$$\min_{K} \|N(K, P)\|_{m}, \ m = 2, \infty$$

- Youla parametrization: search over all stabilizing K(s) ⇒ search over all stable transfer-functions Q(s)
  - the model matching problem
- Linear Matrix Inequalities: translate optimization problem into low-complexity convex problem
- Model reduction: optimization problem typically yields high-order *K*(*s*) → reduce order of controller while maintaining essential properties



# Parametrization of all stabilizing controllers

Internal Model Control (IMC) structure



Assume G stable. Then closed-loop internally stable iff

$$K(I+GK)^{-1} = Q$$
  $(I+GK)^{-1} = I-GQ$   
 $(I+KG)^{-1} = I-QG$   $G(I+KG)^{-1} = G(I-QG)$ 

all stable  $\Leftrightarrow Q$  stable

- The error feedback controller  $K = Q(I GQ)^{-1}$
- Thus, a parametrization of all stabilizing controllers is

$$K = Q(I - GQ)^{-1}$$

where Q(s) is any stable transfer-function matrix



### Example: $\mathcal{H}_{\infty}$ Model Matching Problem

The model matching problem: find stable Q(s) such that

$$\| oldsymbol{P}_1(oldsymbol{s}) - oldsymbol{P}_2(oldsymbol{s}) oldsymbol{Q}(oldsymbol{s}) oldsymbol{P}_3(oldsymbol{s}) \|_\infty < \gamma$$

Example: find controller such that  $||w_P S||_{\infty} < 1$ 

• From IMC: 
$$S = I - GQ$$

• Introduce  $P_1 = w_P$ ,  $P_2 = w_P G$ ,  $P_3 = I$ ,  $\gamma = 1$ . Then

 $\| w_P S \|_{\infty} < 1 \quad \Leftrightarrow \quad \| P_1(s) - P_2(s) Q(s) P_3(s) \|_{\infty} < \gamma$ 

## Parametrization for unstable plants

• Left coprime factorization

$$G(s) = M^{-1}(s)N(s)$$

such that M, N proper, stable and satisfy Bezout identity

NX + MY = I

Parametrization of all stabilizing controllers

 $K(s) = (Y(s) - Q(s)N(s))^{-1}(X(s) + Q(s)M(s))$ 

where Q(s) is any stable transfer-matrix



# Strong Stabilizability

- A plant *G* is strongly stabilizable if there exist a stable controller that stabilizes the plant, i.e., the set of stabilizing controllers include stable K(s).
- *Youla:* a strictly proper SISO plant with an odd number of real RHP poles to the right of any RHP zero is not strongly stabilizable.

Example: The plant

$$G(s) = rac{(s-1)}{(s-2)(s+3)}$$

can only be stabilized by an unstable controller.

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Similar result for MIMO plants



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# Linear Matrix Inequalities

Linear Matrix Inequality (LMI)

$$F_0+F_1x_1+F_2x_2+\ldots F_mx_m>0$$

where

- $-x = [x_1 \dots x_m]$  is a real vector
- $-F_i$ , i = 0, m are symmetric real matrices

An LMI imposes a *convex constraint* on x

- feasibility problem: find some x that satisfies LMI
- linear optimization problem: minimize  $c^T x$  subject to LMI
- generalized eigenvalue problem: minimize largest generalized eigenvalue subject to LMI

#### Old problem, efficient solvers now available

#### Remark: LMIs often written on matrix form also for x

$$F_0 + \sum_{i=1}^n G_i X_i H_i > 0$$

where  $G_i$ ,  $H_i$  are given matrices and we seek  $X_i$ 



# Example 1: Linear stability problem

LTI system

$$\dot{x} = Ax(t)$$

• Lyapunov function  $V(x) = x^T P x > 0$  with  $\dot{V}(x) < 0$  iff

$$P = P^T > 0$$
,  $A^T P + P A < 0$ 

 Corresponds to an LMI feasibility problem (just stack the two inequalities into one big matrix)



# Example 2: Linear robust stability problem

Polytopic LTV system

$$\dot{x} = A(t)x(t), \quad A(t) \in \{A_1, \ldots, A_L\}$$

• Lyapunov function exist iff

$$P = P^T > 0$$
,  $A_i^T P + P A_i < 0$ ,  $i = 1, \dots L$ 

• LMI feasibility problem



# Optimization problems: $\mathcal{H}_{\infty}$ -norm

Consider LTI system

$$\dot{x} = Ax(t) + Bw(t)$$
$$z(t) = Cx(t) + Dw(t)$$

The  $\mathcal{H}_\infty$  norm of  $\textit{G}_{\textit{zw}}$  is equivalent to solving

$$\min \gamma \quad s.t. \quad \begin{pmatrix} A^T P + PA & PB & C^T \\ B^T P & -\gamma I & D^T \\ C & D & -\gamma I \end{pmatrix} < 0, \quad P > 0$$

i.e., minimization subject to LMI.

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# Upper Bound on $\mu_{\Delta}(N)$

Recall upper bound for  $\mu$ 

$$\mu_{\Delta}(N) \leq \min_{D \in \mathcal{D}} \bar{\sigma}(DND^{-1}); \quad \mathcal{D}: D\Delta = \Delta D$$

Translate into LMI

$$\bar{\sigma}(DND^{-1}) \leq \gamma \quad \Leftrightarrow \quad \rho(D^{-H}N^{H}D^{H}DMD^{-1}) < \gamma^{2}$$
$$\Leftrightarrow \quad D^{-H}N^{H}D^{H}DND^{-1} - \gamma^{2}I < 0 \quad \Leftrightarrow \quad N^{H}PN - \gamma^{2}P < 0$$
where  $P = D^{H}D > 0$  and has same structure as  $D$ .

Corresponds to generalized eigenvalue problem

min 
$$\gamma^2$$
 s.t.  $N^H P N - \gamma^2 P < 0$ 

# LMIs in control - the Essence

- Matlab: LMI toolbox, and LMI Lab in Robust Control toolbox.
- See EL3300 Convex Optimization with Engineering Applications



# Todays program

Optimal control problems

$$\min_{K} \|N(K, P)\|_{m}, \ m=2,\infty$$

- Youla parametrization: search over all stabilizing K(s) ⇒ search over all stable transfer-functions Q(s)
  - the model matching problem
- Linear Matrix Inequalities: translate optimization problem into low-complexity convex problem



## The LTI Model Reduction Problem

Given minimal state-space model (A, B, C, D)

$$\dot{x} = Ax(t) + Bu(t)$$
  $x \in \mathbb{R}^n, u \in \mathbb{R}^m$   
 $y(t) = Cx(t) + Du(t)$   $y \in \mathbb{R}^l$ 

or, as input-output model

$$Y(s) = \underbrace{\left[C(sI - A)^{-1}B + D\right]}_{G(s)} U(s)$$

model reduction: we seek a model

$$\dot{x}_r = A_r \hat{x}(t) + B_r u(t)$$
  $x_r \in \mathbb{R}^k, u \in \mathbb{R}^m$   
 $y(t) = C_r x_r(t) + D_r u(t)$   $y \in \mathbb{R}^l$   
 $G_a(s) = C_r (sl - A_r)^{-1} B_r + D_r$ 

with k < n, such that the predicted input-output behavior is close in some sense, e.g.,  $\|G - G_a\|_{\infty}$  is small

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# Why Model Reduction?

- Reduced computational complexity
  - time for dynamic simulation is approximately proportional to n<sup>3</sup> (if A dense)
  - in particular, important for real time applications, e.g., controllers
- Controller synthesis methods typically yield controllers that have order at least equal to model order, usually significantly higher. Thus, to obtain low order controller
  - reduce model order prior to control design, or
  - reduce controller order after design

A multitude of model reduction methods. Here we will consider those most commonly employed in linear control theory.



### Truncation and Residualization

Divide state vector x into two vectors  $x_1$  and  $x_2$  of dimension k and n - k, respectively

$$\dot{x}_1 = A_{11}x_1(t) + A_{12}x_2(t) + B_1u(t)$$
  
$$\dot{x}_2 = A_{21}x_1(t) + A_{22}x_2(t) + B_2u(t)$$
  
$$y(t) = C_1x_1(t) + C_2x_2(t) + Du(t)$$

We aim at removing the state vector  $x_2$ , i.e., obtain a *k*th order model from an *n*th order model

- **Truncation:** let  $x_2 = 0$ , i.e., remove  $x_2$  from state-space model
- **Residualization:** let  $\dot{x}_2 = 0$ , i.e.,  $x_2$  becomes an algebraic variable which depends on  $x_1$  and u



#### Truncation

With  $x_2 = 0$  we get

$$(A_r, B_r, C_r, D_r) = (A_{11}, B_1, C_1, D)$$

- Simply removing a number of states makes little sense in general
- Consider first transforming (A, B, C, D) into Jordan form and arrange the states so that x<sub>2</sub> correspond to the fastest modes
- If the Jordan form is diagonal (distinct eigenvalues  $\lambda_i$ ) then

$$G(s) = \sum_{i=1}^{n} \frac{c_i b_i^T}{s - \lambda_i}$$

Removing the n – k fastest modes then yields the model error

$$G(s) - G_a(s) = \sum_{i=k+1}^n \frac{c_i b_i^{\mathsf{T}}}{s - \lambda_i} \quad \Rightarrow \quad \|G - G_a\|_\infty \leq \sum_{i=k+1}^n \frac{\bar{\sigma}(c_i b_i^{\mathsf{T}})}{|\mathsf{Re}(\lambda_i)|}$$

note: must assume stable G(s)



## Truncation cont'd

• The  $H_{\infty}$  error bound

$$\sum_{i=k+1}^{n} \frac{\bar{\sigma}(\boldsymbol{c}_{i}\boldsymbol{b}_{i}^{T})}{|\boldsymbol{R}\boldsymbol{e}(\lambda_{i})|}$$

depends not only on eigenvalues of fast modes, but also on the residues  $c_i b_i^T$ , i.e., the effect of inputs *u* on  $x_2$  and effect of  $x_2$  on outputs *y* 

• At 
$$\omega = \infty$$

$$G_a(i\infty) = G(i\infty) = D$$

Thus, no error at infinite frequency

### Residualization

With  $\dot{x}_2 = 0$  we get (assume  $A_{22}$  invertible)

$$x_2(t) = -A_{22}^{-1}A_{21}x_1(t) - A_{22}^{-1}B_2u(t)$$

and elimination of  $x_2$  from partitioned model then yields

$$\dot{x}_{1}(t) = (A_{11} - A_{12}A_{22}^{-1}A_{21})x_{1}(t) + (B_{1} - A_{12}A_{22}^{-1}B_{2})u(t)$$
$$y(t) = (C_{1} - C_{2}A_{22}^{-1}A_{21})x_{1}(t) + (D - C_{2}A_{22}^{-1}B_{2})u(t)$$

• Thus, the reduced model  $(A_r, B_r, C_r, D_r) =$ 

$$(A_{11} - A_{12}A_{22}^{-1}A_{21}, B_1 - A_{12}A_{22}^{-1}B_2, C_1 - C_2A_{22}^{-1}A_{21}, D - C_2A_{22}^{-1}B_2)$$

- Corresponds to a *singular perturbation method* if A transformed to Jordan form first
- At zero frequency

$$G_a(0) = G(0)$$

follows from the fact that  $\dot{x}_2 \equiv 0$  at steady-state



## Comments on Truncation and Residualization

- Truncation gives best approximation at high frequencies
- Residualization gives best approximation at low frequencies
- The two methods are related through the bilinear transformation  $s \rightarrow \frac{1}{s}$
- Both methods can in principle give rise to arbitrarily large model reduction errors since effect of states on input-output behavior not only related to the speed of response
- Should be combined with some method that ensures relatively small overall effect of removed states on input-output behavior ⇒ *balancing*



## The Controllability Gramian

The state space model (A, B, C, D) has an impulse response from u(t) to x(t) given by

$$X(t)=e^{At}B$$

• A quantification of the "size" of the impulse response is

$$P(t) = \int_0^t X(\tau) X^{\mathsf{T}}(\tau) d\tau = \int_0^t e^{A\tau} B B e^{A^{\mathsf{T}}\tau} d\tau$$

• Define the Controllability Gramian P as

$$P = \lim_{t \to \infty} P(t)$$

The controllability gramian can be computed from the Lyapunov equation

$$AP + PA^T + BB^T = 0$$

• *P* is a quantitative measure for controllability of the different states. Essentially, measures the effect of the inputs on the different states



## The Observability Gramian

The state space model (A, B, C, D) with input u(t) = 0 and initial state x(0) = x\* has the output

$$y(t) = Ce^{At}x^*$$

The energy of the output

$$\int_0^t y^T(\tau) y(\tau) d\tau = x^{T*} \underbrace{\int_0^t e^{A^T \tau} C^T C e^{A\tau} d\tau}_{t} x^{*}$$

• Define the Observability Gramian Q as

$$Q = \lim_{t \to \infty} \int_0^t e^{A^{\tau} \tau} C^{\tau} C e^{A \tau} d\tau$$

Q(t)

• The observability gramian can be computed from the Lyapunov equation

$$A^T Q + Q A + C^T C = 0$$

Q is a quantitative measure for observability of the different states.
Essentially measures the effect of states on outputs
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## **Balanced Realizations**

 We seek a similarity transformation of the states x<sub>b</sub>(t) = Tx(t) so that the transformed state space model

$$\dot{x}_b = TAT^{-1}x_b(t) + TBu(t)$$
$$y(t) = CT^{-1}x_b(t) + Du(t)$$

has controllability and observability gramians

$$P = Q = diag(\sigma_1, \ldots, \sigma_n); \quad \sigma_i = \sqrt{\lambda_i(PQ)}$$

where the Hankel singular values  $\sigma_1 > \sigma_2 > \ldots > \sigma_n$ 

- Each state x<sub>bi</sub> in the balanced realization is as observable as it is controllable, and σ<sub>i</sub> is a measure of how controllable/observable it is
- A state with a relatively small σ<sub>i</sub> has a relatively small effect on the input-output behavior and can hence be removed without significantly affecting the predicted input-output behavior

### Balanced Truncation and Residualization

• Consider the balanced realization (A, B, C, D) of G(s) with partitioning

$$A = \begin{pmatrix} A_{11} & A_{12} \\ A_{21} & A_{22} \end{pmatrix}, \quad B = \begin{pmatrix} B_1 \\ B_2 \end{pmatrix}, \quad C = \begin{pmatrix} C_1 & C_2 \end{pmatrix}$$
$$P = Q = \begin{pmatrix} \Sigma_1 & 0 \\ 0 & \Sigma_2 \end{pmatrix}$$

where  $\Sigma_1 = diag(\sigma_1, \ldots, \sigma_k)$  and  $\Sigma_2 = diag(\sigma_{k+1}, \ldots, \sigma_n)$ 

 A balanced truncation or residualization retaining the k states corresponding to Σ<sub>1</sub> will both have model reduction error

$$\|\boldsymbol{G}-\boldsymbol{G}_{\boldsymbol{a}}^{k}\|_{\infty}\leq 2\sum_{i=k+1}^{''}\sigma_{i}$$

"twice the sum of the tail"

 May in principle include frequency dependent weighting to emphasize certain frequency ranges, However, this introduces extra states and it is furthermore usually non-trivial to choose weigths that give the desired result

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## **Optimal Hankel Norm Approximation**

• The Hankel norm of a transfer-matrix E(s)

$$\|\boldsymbol{E}(\boldsymbol{s})\|_{\boldsymbol{H}} = \sigma_1 = \sqrt{\rho(\boldsymbol{P}\boldsymbol{Q})}$$

i.e., equals the maximum Hankel singular value of E(s)

- Optimal Hankel norm approximations seeks to minimize ||G G<sup>k</sup><sub>a</sub>||<sub>H</sub> for a given order k of the reduced order model
- For stable square *G*(*s*) the optimal Hankel norm *k*th order approximation can be directly computed and has Hankel norm error

$$\|\boldsymbol{G}-\boldsymbol{G}_a^k\|_H=\sigma_{k+1}$$

The optimal Hankel norm is independent of the *D*-matrix of G<sup>k</sup><sub>a</sub> The minimum ∞-norm of the error is

$$\min_{D} \|\boldsymbol{G} - \boldsymbol{G}_{\boldsymbol{a}}^{k}\|_{\infty} \leq \sum_{i=k+1}^{n} \sigma_{i}$$

i.e., "sum of the tails" only

## Unstable models

Balanced truncation and residualization and optimal Hankel norm approximations applies to stable G(s) only. Two "tricks" to deal with unstable models

Separate out unstable part before performing model reduction of stable part

$$G(s) = G_u(s) + G_s(s) \quad \Rightarrow \quad G_a(s) = G_u(s) + G_{sa}(s)$$

2 Consider coprime factorization of G(s)

$$G(s) = M^{-1}(s)N(s)$$

with M(s) and N(s) stable. Apply model reduction to [M(s) N(s)] and use

$$G_a(s) = M_a^{-1}(s)N_a(s)$$

## Model Reduction in Matlab

- modreal: truncation or residualization
- slowfast: slow/fast mode decomposition
- balreal: balanced realization
- hankelmr: optimal Hankel norm approximation
- stabproj: decompose into stable and antistable parts
- ncfmr: balanced model truncation for normalized coprime factors



## Model Reduction Course

#### • EL3500 Introduction to Model Order Reduction



## Learning Outcomes

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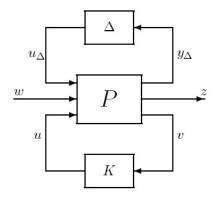


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# Learning Outcomes





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## The Final Moment

#### Exam:

- Covers Lectures and Ch. 1-9 (+ Ch. 11-12 tutorial) in Skogestad and Postlethwaite
- 1-day take home exam, open book.
  - allowed aids: course book(s), lecture slides, matlab, calculator
  - not allowed: exercises, solutions, or anything but the above
- available between December 15 and January 15
- send an email to jacobsen@kth.se with the date at which you want to take it (give at least 2 days notice)
- all exercises must be approved prior to taking out exam

